

## **ARCADIA aeroacoustic design code**

Sergio Campobasso

Mihai Duta

Mike Giles

Lorenzo Lafronza

Alistair Laird

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## Overview

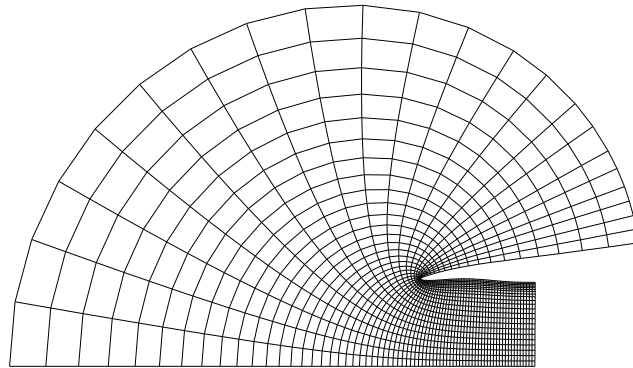
Oxford role is to contribute expertise in CFD and the use of adjoints for design sensitivities through two codes:

- HYDRA – existing Euler/Navier-Stokes code for a range of applications
- ARCADIA – new potential flow code for fan tone noise

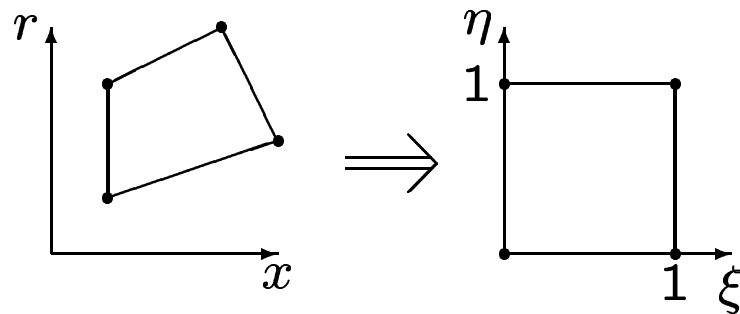
This presentation will focus on ARCADIA, and its novel use of asymptotics and adjoint methods.

## Axisymmetric Geometry

2D axisymmetric grid



with each cell mapped to a unit square



## Axisymmetric Geometry

Coordinates and potential flow solution are interpolated from nodal values:

$$\begin{aligned}x(\xi, \eta) &= \sum_n x_n N_n(\xi, \eta), \\r(\xi, \eta) &= \sum_n r_n N_n(\xi, \eta), \\\phi(\xi, \eta) &= \sum_n \phi_n N_n(\xi, \eta), \\\hat{\phi}(\xi, \eta) &= \sum_n \hat{\phi}_n N_n(\xi, \eta),\end{aligned}$$

Putting this into the weak form of the potential flow equations gives discrete systems of equations for the nodal values.

## Axisymmetric Geometry

Steady flow analysis leads to a coupled system of nonlinear equations

$$R(\phi) = 0$$

which is solved by Newton iteration

$$K_\nu \Delta \phi_\nu = -R_\nu$$

where  $K_\nu = \left( \frac{\partial R}{\partial \phi} \right)_\nu$ .

## Axisymmetric Geometry

The unsteady flow analysis for a prescribed circumferential mode number

$$\exp(i\omega t + i\kappa\theta) \hat{\phi}(x, r)$$

leads to a linear system of equations

$$(-\omega^2 M + i\omega C + K + \kappa^2 K^\theta) \hat{\phi} = \hat{f}$$

with the forcing term  $\hat{f}$  coming from the modal boundary condition on the fan face.

Currently, ARCADIA does not have an acoustic liner model, and the far-field non-reflecting boundary conditions are low-order.

## Non-axisymmetric Geometry

Three ways of handling non-axisymmetry:

- 3D finite elements, with hexahedral elements mapped to unit cubes

$$\phi(\xi, \eta, \zeta) = \sum_n \phi_n N_n(\xi, \eta, \zeta)$$

- spectral elements with  $x, r, \phi, \hat{\phi}$  expressed as Fourier series

$$\phi(\xi, \eta, \theta) = \sum_{m,n} \phi_{m,n} \exp(im\theta) N_n(\xi, \eta)$$

- asymptotic analysis, like the spectral approach but assuming small asymmetry

## Non-axisymmetric Geometry

Two kinds of error:

- 3D – spectral =  
error due to circumferential resolution
- asymptotic – spectral =  
error due to linearisation

Pros and cons:

- 3D analysis straightforward in principle,  
but needs a fine grid so costly in practice
- spectral elements require many fewer  
unknowns, but tricky to solve the equations
- asymptotic analysis uses 2D calculations so  
cheap, but can it handle large perturbations?

## Non-axisymmetric Geometry

Steps in asymptotic analysis:

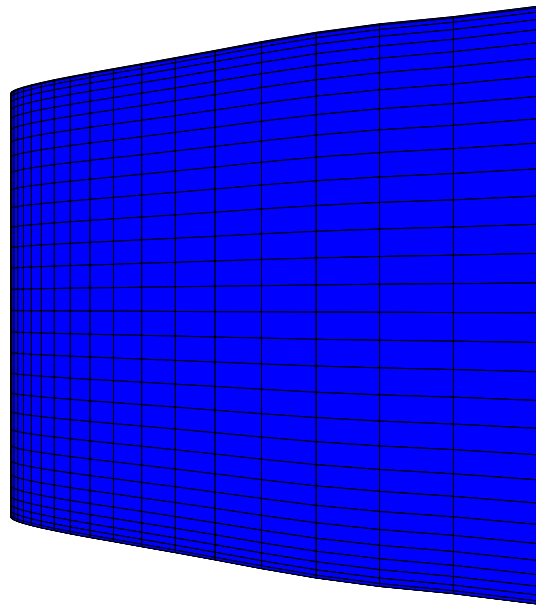
- start with decomposition of geometry into axisymmetric average plus perturbation
- compute  $\phi_0$ , axisymmetric average flow field
- solve a linear perturbation equation for each circumferential flow field perturbation mode

$$L_m \phi_m + A_m x_m + B_m r_m = 0,$$

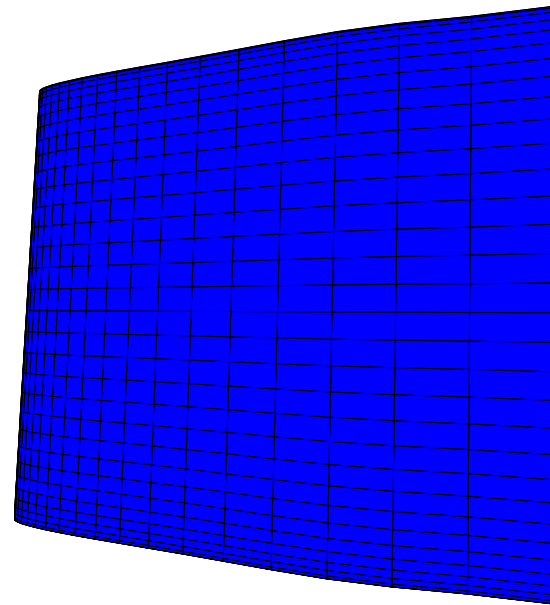
- compute  $\hat{\phi}_0$ , axisymmetric part of linear unsteady solution
- solve separate equations for the other modes

$$\hat{L}_m \hat{\phi}_m + \hat{A}_m x_m + \hat{B}_m r_m + \hat{C}_m \phi_m = 0$$

## Steady validation

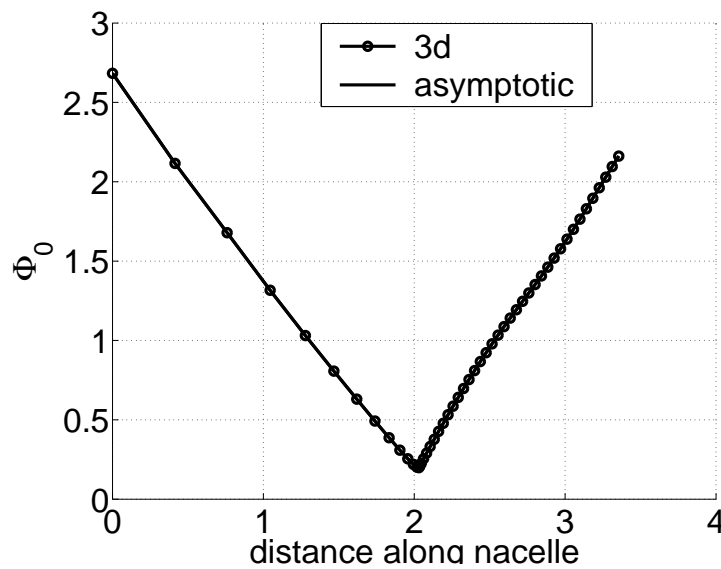


Axisymmetric

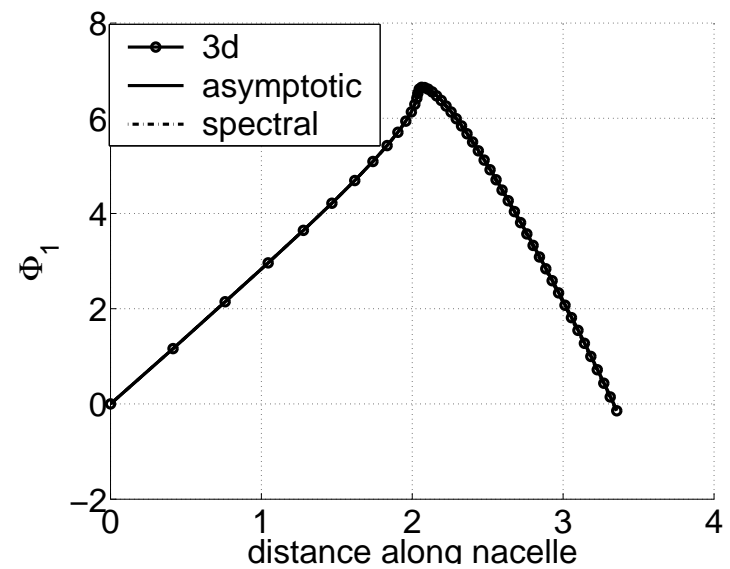


Asymmetric

## Steady validation

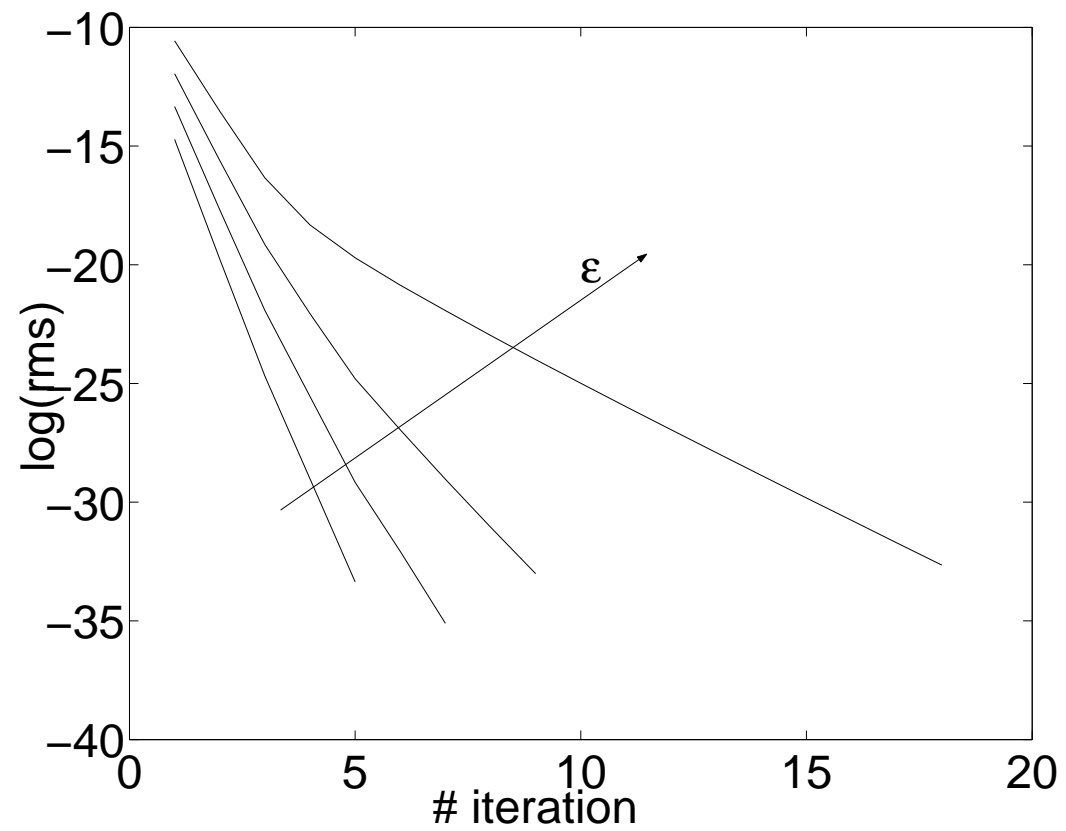


0th harmonic



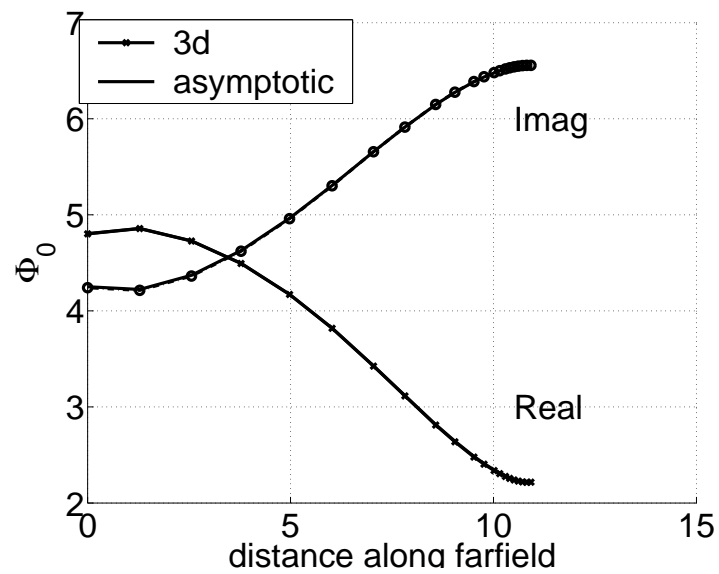
1st harmonic

### Steady validation

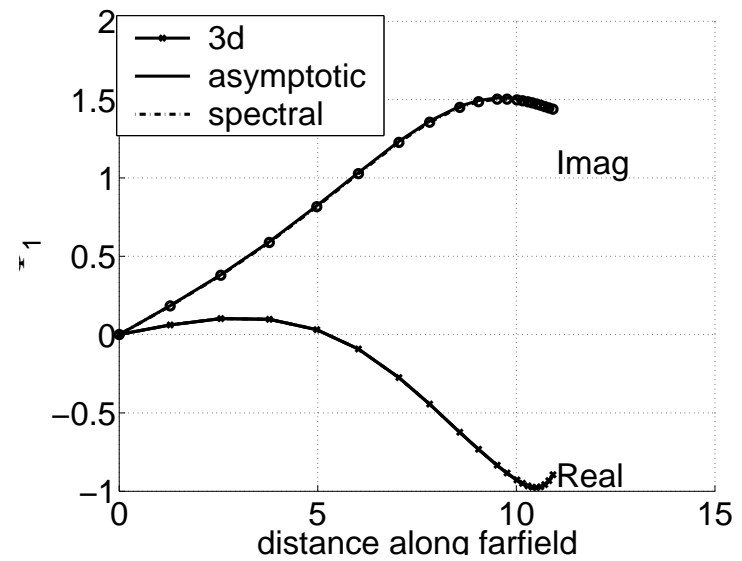


iterative convergence of spectral solver

## Unsteady validation



0th harmonic



1st harmonic

## Adjoint Equations

For some output functional  $J(x, r, \phi)$  the sensitivity to changes in a design variable  $\alpha$  is

$$\frac{dJ}{d\alpha} = \sum_m \left( \frac{\partial J}{\partial x_m} \frac{\partial x_m}{\partial \alpha} + \frac{\partial J}{\partial r_m} \frac{\partial r_m}{\partial \alpha} + \frac{\partial J}{\partial \phi_m} \frac{\partial \phi_m}{\partial \alpha} \right)$$

The flow sensitivity is given by

$$L_m \frac{\partial \phi_m}{\partial \alpha} + A_m \frac{\partial x_m}{\partial \alpha} + B_m \frac{\partial r_m}{\partial \alpha} = 0,$$

## Adjoint Equations

Hence, by defining the adjoint variables  $v_m$  by the equation

$$L_m^H v_m = - \left( \frac{\partial J}{\partial \phi_m} \right)^H ,$$

one obtains

$$\begin{aligned} \frac{dJ}{d\alpha} = \sum_m \left\{ \left( v_m^H A_m + \frac{\partial J}{\partial x_m} \right) \frac{\partial x_m}{\partial \alpha} \right. \\ \left. + \left( v_m^H B_m + \frac{\partial J}{\partial r_m} \right) \frac{\partial r_m}{\partial \alpha} \right\} . \end{aligned}$$

This gives a very efficient way of computing the sensitivity to multiple design variables.

## Adjoint Equations

For functionals  $\hat{J}(x, r, \phi, \hat{\phi})$  depending on the unsteady flow field, there is a two-stage process.

First one solves the unsteady adjoint equations

$$\hat{L}_m^H \hat{v}_m = - \left( \frac{\partial \hat{J}}{\partial \hat{\phi}_m} \right)^H$$

and then one solves the steady adjoint equations

$$L_m^H v_m = - \left( \hat{v}_m^H \hat{C}_m + \frac{\partial \hat{J}}{\partial \phi_m} \right)^H$$

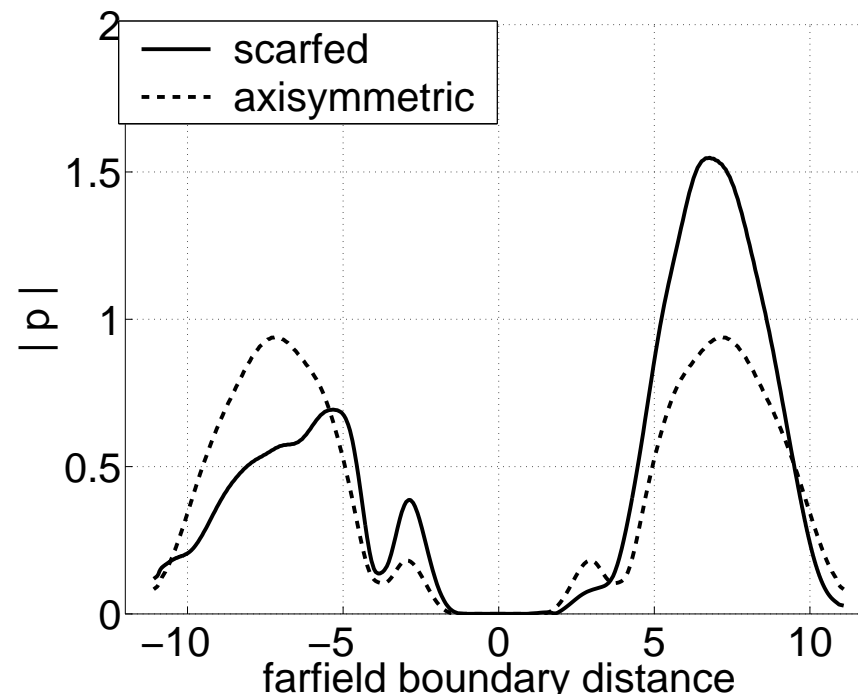
## Adjoint Equations

The final sensitivity result is then

$$\frac{d\hat{J}}{d\alpha} = \sum_m \left\{ \left( v_m^H A_m + \hat{v}_m^H \hat{A}_m + \frac{\partial \hat{J}}{\partial x_m} \right) \frac{\partial x_m}{\partial \alpha} + \left( v_m^H B_m + \hat{v}_m^H \hat{B}_m + \frac{\partial \hat{J}}{\partial r_m} \right) \frac{\partial r_m}{\partial \alpha} \right\}.$$

Because we explicitly assemble all of the necessary matrices for the asymptotic analysis, the implementation of the adjoint analysis is trivial. This will be used with gradient-based optimisation in GEODISE.

## Objective



Reduce the noise radiated towards the ground by re-designing the nacelle (negative scarfing).